## Introduction to Aerospace Engineering

Lecture slides







Faculty of Aerospace Engineering 29-11-2011





# Learning objectives

#### Student should be able to...

- Describe the most relevant loads for
  - An aircraft
  - A spacecraft
- Explain whether these loads are
  - Static or dynamic
  - Concentrated or distributed



#### Aircraft structure

• The airframe is externally loaded





#### Aircraft structure

The airframe is externally loaded

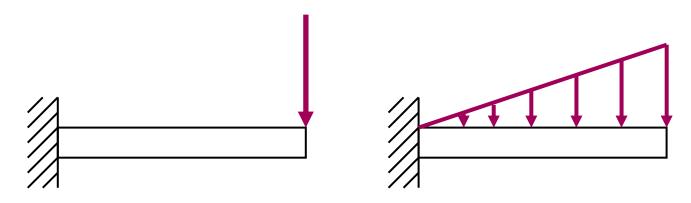




#### Aircraft structure

- The airframe is externally loaded
  - Maneuvers
  - Gust
  - Cabin pressure
  - Landing
  - etc.

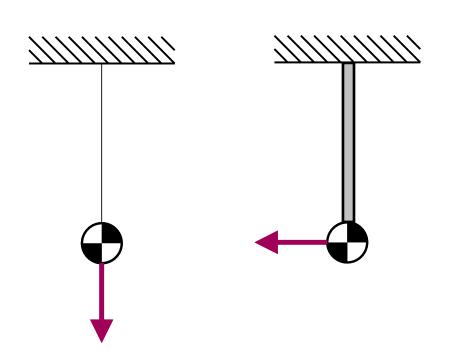
by concentrated or distributed forces

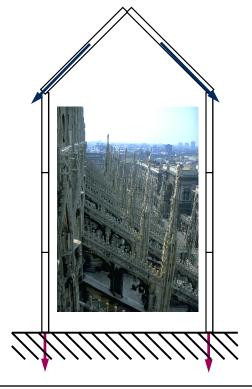




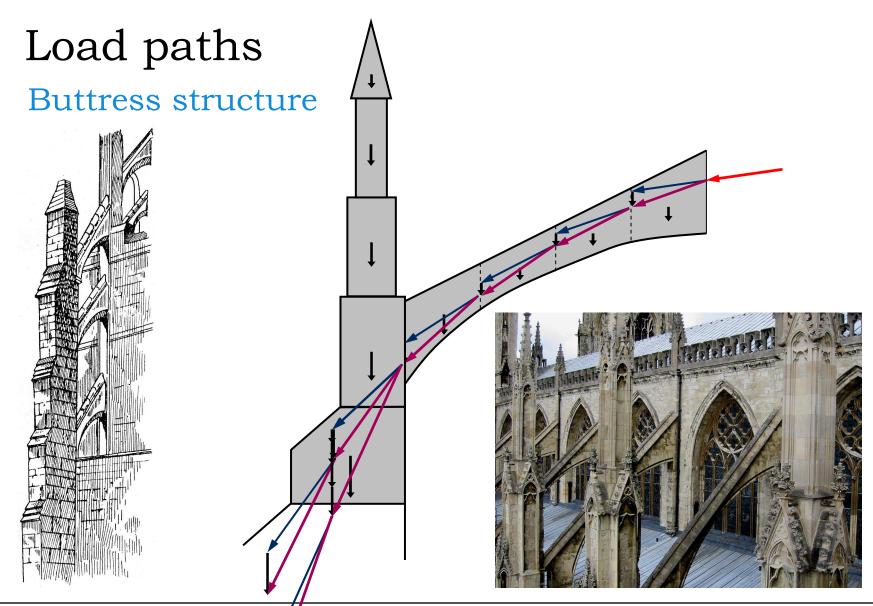
### Simple structure

- What is a load path?
  - Path to link applied load to equilibrium forces



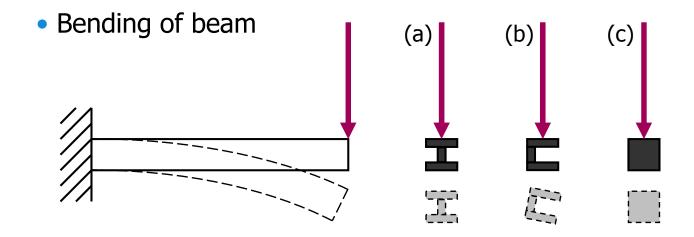








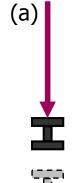
### Simple structure

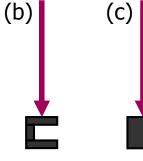


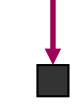


### Simple structure

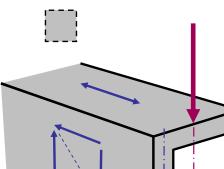
Bending of beam











Load path

Upper girder: tension

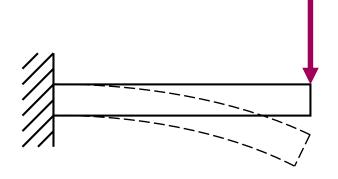
Lower girder: compression

Web plate: shear



### Simple structure

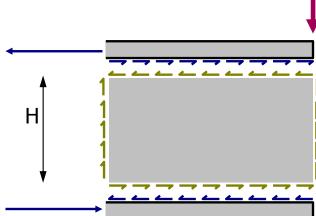
Bending of beam



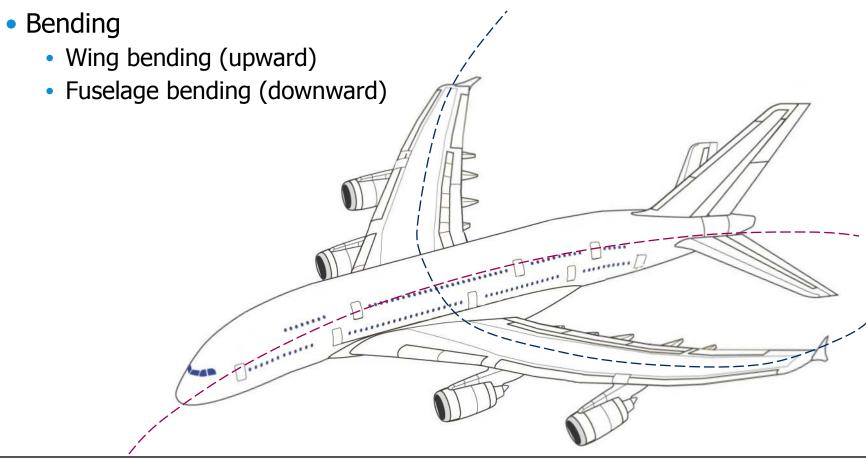




- Load path
  - Upper girder: tension
  - Lower girder: compression
  - Web plate: shear

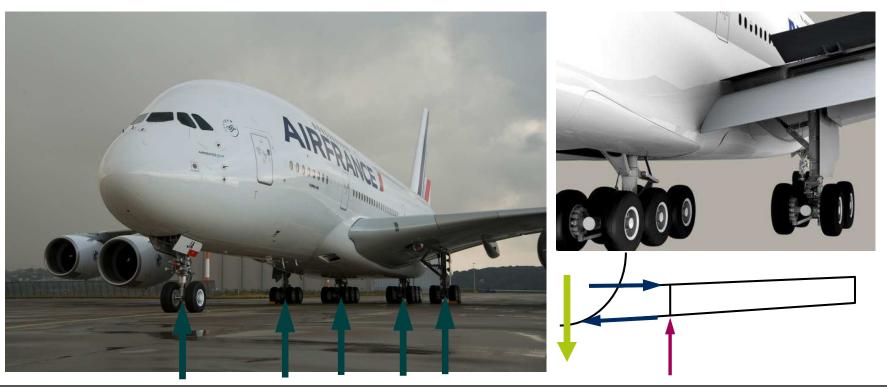






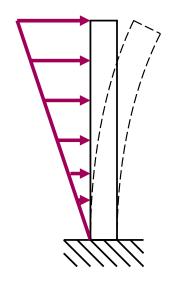


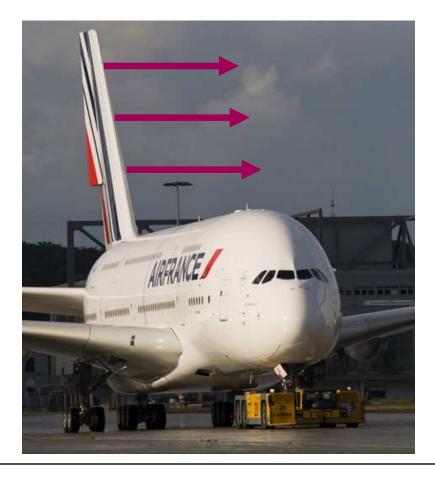
- Landing and taxiing
  - 'concentrated forces' acting on undercarriage





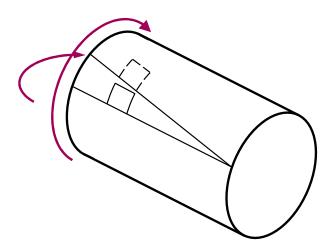
- Vertical tail load
  - Bending of vertical tail

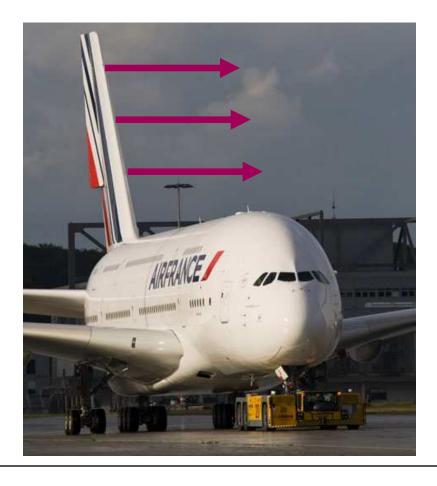






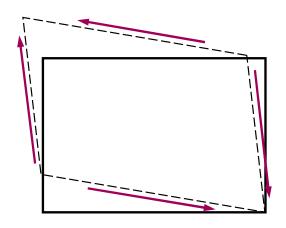
- Vertical tail load
  - Bending of vertical tail
  - Bending & rotation of fuselage

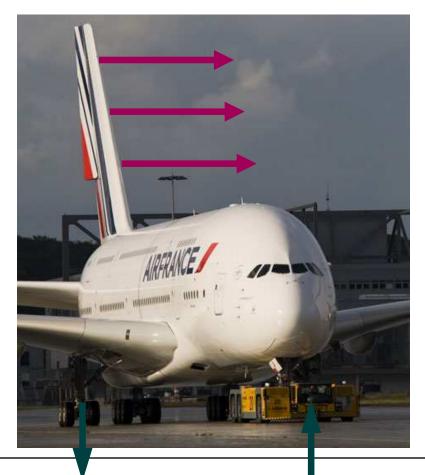






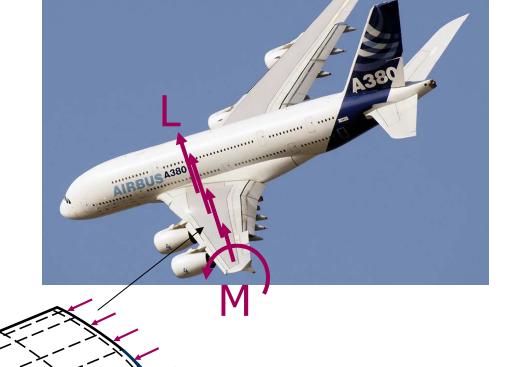
- Vertical tail load
  - Bending of vertical tail
  - Bending & rotation of fuselage
  - Shear of fuselage side panels







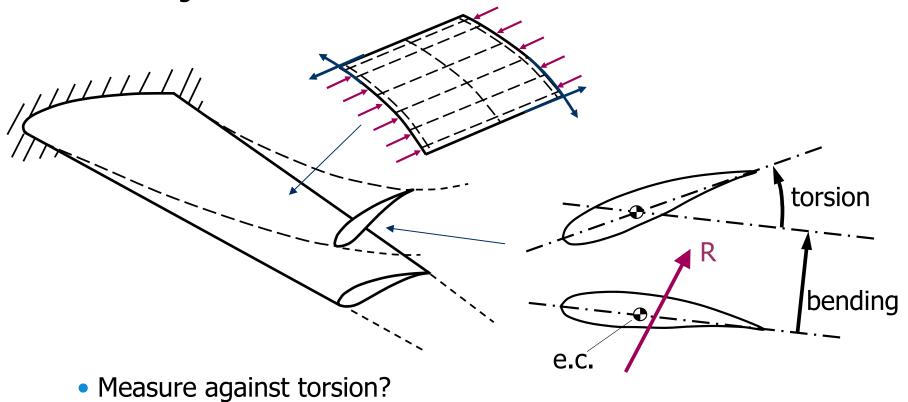
- Wing load ⇒ deformation
  - Upward bending of wing
  - Rotation of wing
- Deformation ⇒ stresses
  - Compression and shear in upper wing skin panel
  - Tension and shear in lower wing skin panel





#### Airframe

Bending and torsion





17 | 26

### Load introduction

Wing/fuselage connection

- Example
  - A400M





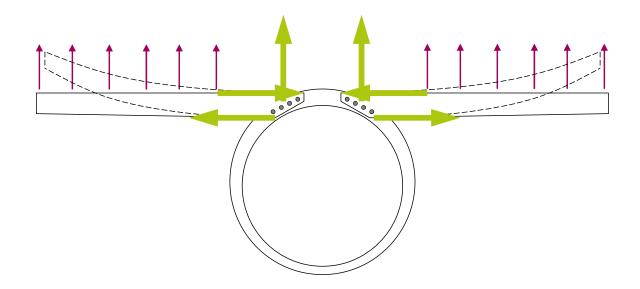




### Load introduction

### Wing/fuselage connection

- Example
  - A400M

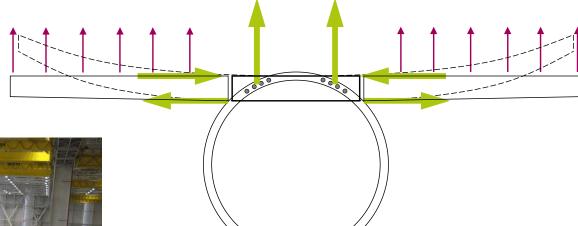




### Load introduction

### Wing/fuselage connection

- Example
  - A400M





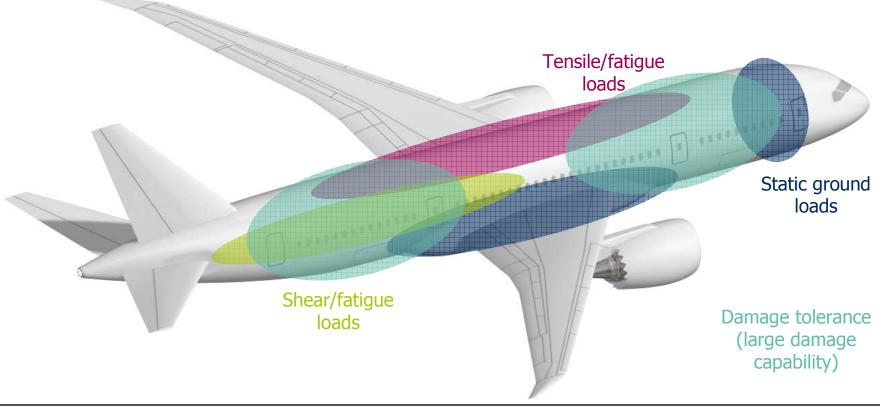
Wing/fuselage connection F-27
See question 14 related to the study collection



## Design load cases

### Aircraft fuselage structure

Different locations ⇒ different loads & criteria!





# Design load cases

#### Aircraft structure

- Dynamic loads (example of flutter)
  - Oscillation of aircraft component caused by the interaction of aerodynamic forces, structural elastic reactions, and inertia





#### Spacecraft structure

- Gravity
  - Handling and transportation loads
- Vibration and acoustic test loads
- Launch loads (static & dynamic)
  - Quasi-static
  - Sine vibration
  - Acoustic noise and random vibration
  - Shock loads
- In-orbit loads
  - Shocks
  - Structurally transmitted loads
  - Internal pressure
  - Thermal stress





### Spacecraft structure

Steady state load

Axial: launch vehicle engine trust

Lateral: wind gust & vehicle maneuvers



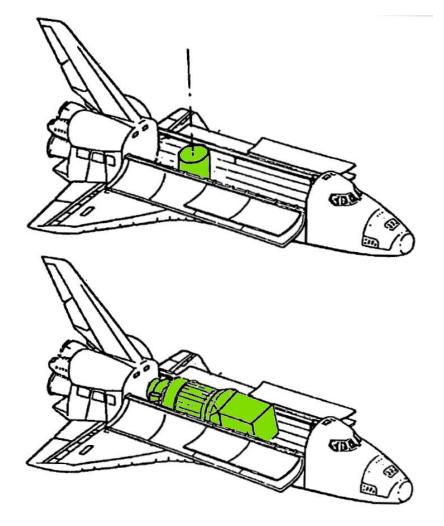


### Spacecraft structure

- Load path



 Primary axis // direction of maximum acceleration





### Summary

### Aircraft & spacecraft loads

- Identify relevant loads
  - Static or dynamic
  - Concentrated or distributed
- Load paths
  - Load introduction
  - Equilibrium

