AE4520: Advanced Structural Analysis Take Home Exam

Due date January 16th 2014

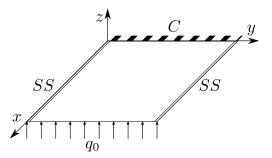


Figure 1: Rectangular Plate under Edge Load.

- 1. (30 points) The square plate shown in figure 1 has a side length a and a constant thickness t. The plate is made of an isotropic material having Young's modulus E and Poisson's ratio $\nu = 0.3$. The plate is simply supported (SS) on two opposite sides. One of the two others is clamped (C) and the other is free (F). The plate is subject to a uniform edge line load q_0 .
 - (a) Assume that the deflection is expanded in a series of the form

$$w = \sum_{i} \phi_n(x) \sin \frac{n\pi y}{a}$$

Show, by integrating only over y, that the total potential energy can be written in the form,

$$\Pi = \sum_{n} \Pi_{n},$$

where Π_n depends only on $\phi_n(x)$.

- (b) Using the principle of minimum potential energy find the governing fourth order ordinary differential equations for ϕ_n and their boundary conditions.
- (c) Plot the deflection of the plate indicating maximum displacement.
- (d) Plot the variation of the von Mises stress over the plate and indicate the critical areas.
- 2. (40 points) In modern aerospace structures, there is a relentless drive to reduce weight and cost. One concept that is currently under research is the use of iso-grid stiffened structures. Such structures are constructed from a triangular grid of stiffeners supporting a thin skin. While most of the loads are taken by the supporting grid, the skin is also non-negligibly loaded and need to be properly designed. One possible failure mode for the skin is buckling.

A single traingular cell is modelled as a simply supported equilateral triangular plate of side length a. The notation for the plate is shown in figure 2.

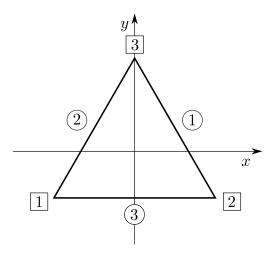


Figure 2: Triangular Panel.

(a) Show that the x and y locations of a general point in the triangle is given by,

$$x = \sum_{i=1}^{3} x_i \xi_i$$
, and, $y = \sum_{i=1}^{3} y_i \xi_i$,

where x_i and y_i are the coordinates of the *i*th vertex and $0 \le \xi_i \le 1$, i = 1...3 are barycentric coordinates satisfying,

$$\xi_1 + \xi_2 + \xi_3 = 1.$$

Find the corresponding equations for ξ_i in terms of the x and y coordinates.

(b) Show that functions of the form,

$$\phi_i = \xi_1^{p_{1i}} \xi_2^{p_{2i}} \xi_3^{p_{3i}}$$

satisfy the simply supported conditions for all integer $p_{1i}, p_{2i}, p_{3i} > 1$.

(c) Assume a Rayleigh-Ritz series of the form,

$$w = \sum_{i} c_i \phi_i$$

containing all the terms such that $p_{1i} + p_{2i} + p_{3i} = p$. Where $p \ge 3$ is the polynomial degree of the approximation.

Use the Rayleigh-Ritz method to find the first four buckling modes and buckling loads under equal biaxial compression. Use Poisson's ratio $\nu = 0.3$. Obtain results for p = 3, 4, 5, 6, and comment on convergence.

[Hint: All derivatives with respect to x and y may be converted to derivatives with respect to the barycentric coordinates. Integrations can be reduced to simpler integrations using the $\xi_1\xi_2$ plane. You might want to use a symbolic manipulator, e.g., maple, to calculate the integrals.]