

opgave no.

naam

studienummer

vak D&S

code AE3-914

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fac.

Gebruik voor elke opgave een afzonderlijk vel papier!

Problem 1

$$\underline{F} \text{ is conservative} \Rightarrow \underline{F} = -\nabla V$$

$$\begin{aligned} \text{consequently } \nabla \times \underline{F} &= -\nabla \times \nabla V = \\ &= -(\nabla \times \nabla)V = \underline{0} \end{aligned}$$

Alternative (analytically):

$$\underline{F} = F_x \underline{i} + F_y \underline{j} + F_z \underline{k} =$$

$$= -\frac{\partial V}{\partial x} \underline{i} - \frac{\partial V}{\partial y} \underline{j} - \frac{\partial V}{\partial z} \underline{k}$$

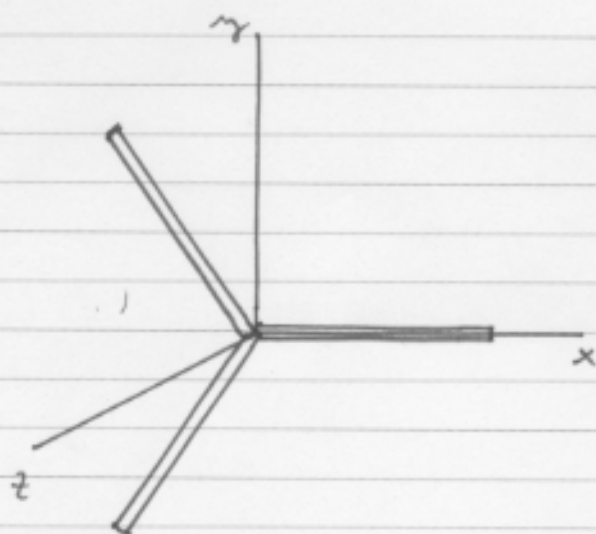
$$\nabla \times \underline{F} = \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ \frac{\partial}{\partial x} & \frac{\partial}{\partial y} & \frac{\partial}{\partial z} \\ -\frac{\partial V}{\partial x} & -\frac{\partial V}{\partial y} & -\frac{\partial V}{\partial z} \end{vmatrix}$$

$$= \frac{\partial^2}{\partial y \partial z} (V-V) \underline{i} + \frac{\partial^2}{\partial x \partial z} (V-V) \underline{j} + \frac{\partial^2}{\partial x \partial y} (V-V) \underline{k}$$

$$= 0 \underline{i} + 0 \underline{j} + 0 \underline{k} = \underline{0}$$

Problem 2

a)



$z \equiv \text{rotor shaft}$

$$I_3 = I_{zz} = 3I_b$$

In xy -plane there are three symmetry axes, consequently all directions are principal in that plane and thus:

$$I_{xx} = I_{yy} = I_1 = I_2 \quad (1)$$

Moreover, the object is contained in the xy -plane, consequently:

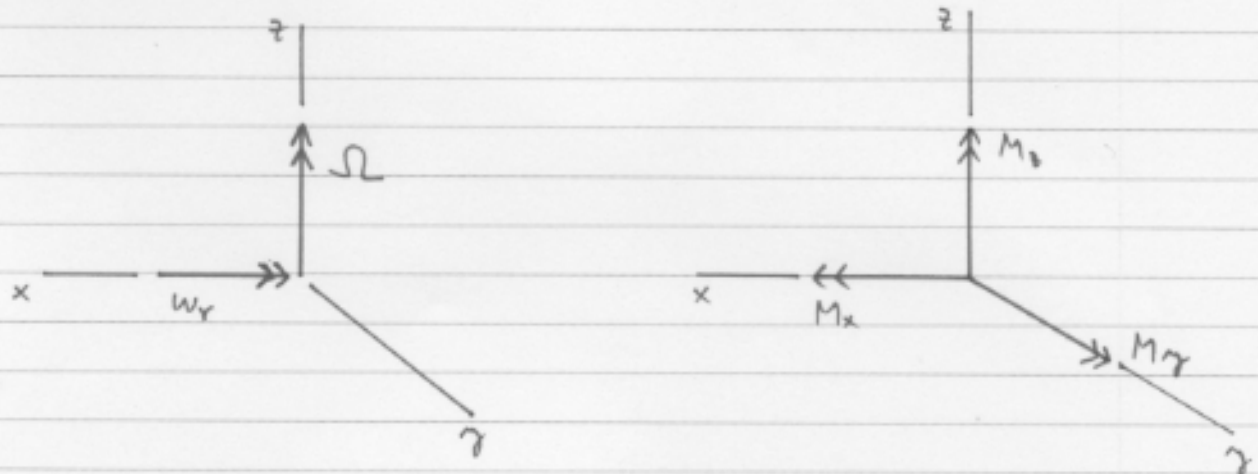
$$\begin{aligned} I_{zz} &= \int (x^2 + y^2) dm = \int x^2 dm + \int y^2 dm \\ &= I_{yy} + I_{xx} \quad (2) \end{aligned}$$

(1) and (2) together provide

$$I_{xx} = I_{yy} = \frac{I_{zz}}{2}; \quad I_1 = I_2 = \frac{I_3}{2}$$

$$I_1 = \frac{3}{2} I_b; \quad I_2 = \frac{3}{2} I_b; \quad I_3 = 3I_b$$

⑥ Forward rotor:



Euler's equations:

$$I_1 \dot{\omega}_x - (I_2 - I_3) \omega_y \omega_z = M_x$$

$$I_2 \dot{\omega}_y - (I_3 - I_1) \omega_x \omega_z = M_y$$

$$I_3 \dot{\omega}_z - (I_1 - I_2) \omega_x \omega_y = M_z$$

Data: $\omega_x = -\omega_r$; $\omega_y = 0$; $\omega_z = \Omega$

$$\dot{\omega}_x = ?; \dot{\omega}_y = ?; \dot{\omega}_z = ?$$

$$\underline{\omega}_{xy} = -\omega_r \underline{i}; \quad \frac{d\underline{\omega}_{xy}}{dt} = \underline{0}$$

$$\frac{d\underline{\omega}_{xy}}{dt} = \dot{\omega}_x \underline{i} + \dot{\omega}_y \underline{j} + \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ \omega_x & \omega_y & \omega_z \\ \omega_x & \omega_y & 0 \end{vmatrix} = \underline{0}$$

$$\Rightarrow \dot{\omega}_x = \omega_y \omega_z; \quad \dot{\omega}_y = -\omega_x \omega_z$$

$$M_x = I_1 \cdot 0 - (I_2 - I_3) \cdot 0 \cdot \Omega = 0$$

$$M_y = I_2 (-\omega_r \Omega) - (I_3 - I_1) (-\omega_r) \Omega = I_3 \omega_r \Omega$$

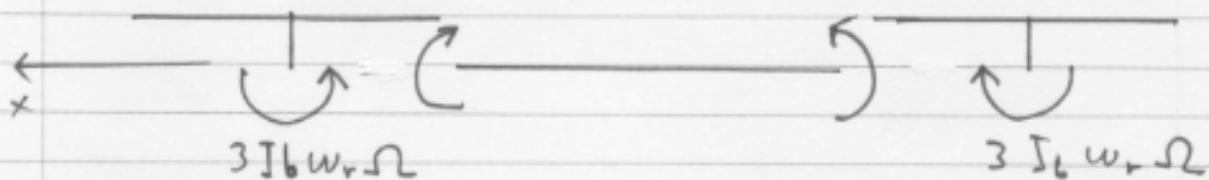
$$= 3 I_b \omega_r \Omega$$

Aft rotor:

Exactly the same as for the forward rotor,
but Ω is now replaced by $-\Omega$

$$M_y = -3 I_b \omega_r \Omega$$

Helicopter:



The bending moment amounts to

$$+ 3 I_b \omega_r \Omega$$

Problem 3

(a)

$$I_f = 2 \int_0^b f^2 dA = 2 \int_0^b f^2 \pm \sqrt{1+f'^2} dy$$
$$= 2 \pm \int_0^b f^2 \sqrt{1+f'^2} dy$$

Euler-Lagrange equation:

For $I = \int_0^b F(y, f, f') dy$ we have

$$\frac{\partial F}{\partial f} - \frac{d}{dy} \left(\frac{\partial F}{\partial f'} \right) = 0 \quad (1)$$

$$\frac{\partial F}{\partial f} = 2 f \sqrt{1+f'^2}$$

$$\frac{\partial F}{\partial f'} = f^2 \frac{f'}{\sqrt{1+f'^2}}$$

$$\frac{d}{dy} \left(\frac{\partial F}{\partial f'} \right) = \frac{2 f f'^2 + f^2 f'' + 2 f f'^3}{(1+f'^2)^{3/2}}$$

Substituting in (1) and rearranging yields

$$2 + 2 f'^2 - f f'' = 0$$

(b)

$$\text{Vstf } 0 = \delta I = \int_0^b \delta F(y, f, f') dy$$

$$= \int_0^b \left[\frac{\partial F}{\partial f} \delta f + \frac{\partial F}{\partial f'} \delta f' \right] dy$$

$$= \int_0^b \frac{\partial F}{\partial f} \delta f dy + \left. \frac{\partial F}{\partial f'} \delta f \right|_0^b$$

$$- \int_0^b \frac{d}{dy} \left(\frac{\partial F}{\partial f'} \right) \delta f dy \quad \Rightarrow$$

$$\text{Vstf } \left. \frac{\partial F}{\partial f'} \delta f \right|_0^b = 0 \Rightarrow \left. \frac{\partial F}{\partial f'} \right|_0 \delta f(0) = 0 \quad (1)$$

$$\left. \frac{\partial F}{\partial f'} \right|_b \delta f(b) = 0 \quad (2)$$

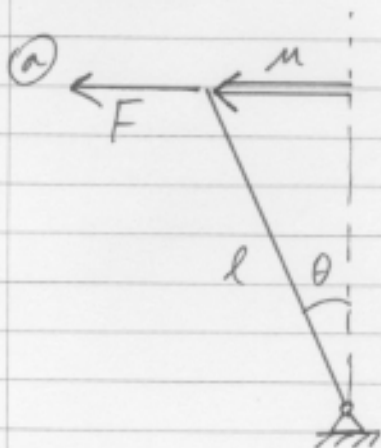
(1) is always 0 because $f(0) = \frac{h}{2} \Rightarrow \delta f(0) = 0$

(2): $f(b)$ is not given, thus $\left. \frac{\partial F}{\partial f'} \right|_b = 0 \rightarrow$ natural boundary condition

$$\frac{\partial F}{\partial f'} = \frac{f^2 f'}{\sqrt{1+f'^2}} = 0 \Rightarrow \left\{ \begin{array}{l} f=0 \\ f'=0 \end{array} \right\} \Rightarrow$$

$$\boxed{\begin{array}{l} f(b) = 0 \\ \text{or} \\ f'(b) = 0 \end{array}}$$

Problem 4



$$u = l \sin \theta; \quad \delta u = l \cos \theta \delta \theta$$

$$\delta W = F \delta u = \underbrace{Fl \cos \theta}_{Q_\theta} \delta \theta$$

$$Q_\theta = Fl \cos \theta$$

which is a moment

$$(b) \quad T = \frac{1}{2} I \dot{\theta}^2 = \frac{1}{6} ml^2 \dot{\theta}^2; \quad V = \frac{1}{2} mgl \cos \theta;$$

$$L = T - V = \frac{1}{6} ml^2 \dot{\theta}^2 - \frac{1}{2} mgl \cos \theta$$

Equation of motion:

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) - \frac{\partial L}{\partial \theta} = Q_\theta$$

$$\frac{1}{3} ml \ddot{\theta} - \frac{1}{2} mg \sin \theta - F \cos \theta = 0$$

$$(c) \quad \theta = 0; \quad \ddot{\theta} = 0$$

$$\frac{1}{3} ml \cdot 0 - \frac{1}{2} mg \sin 0 - F(0) \cos 0 = 0$$

$$\boxed{F(0) = 0}$$

(d)

Convert to first order:

$$\left. \begin{aligned} \dot{\theta} &= \varphi \\ \dot{\varphi} &= \frac{3g \sin \theta}{2l} + \frac{3F \cos \theta}{ml} \end{aligned} \right\}$$

Linearization for $\theta = 0$

$$\begin{bmatrix} \dot{\theta} \\ \dot{\varphi} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{3g \cos 0}{2l} + \frac{3F'(0) \cos 0 - F(0) \sin 0}{ml} & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \varphi \end{bmatrix}$$

$$= \begin{bmatrix} 0 & 1 \\ \frac{3g}{2l} + \frac{3F'(0)}{ml} & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \varphi \end{bmatrix}$$

Eigenvalues:

$$\begin{vmatrix} -\lambda & 1 \\ \frac{3}{l} \left(\frac{g}{2} + \frac{F'(0)}{m} \right) & -\lambda \end{vmatrix} = \lambda^2 - \frac{3}{l} \left(\frac{g}{2} + \frac{F'(0)}{m} \right) = 0$$

Stable if $\lambda \in \text{Im} \Rightarrow \frac{g}{2} + \frac{F'(0)}{m} < 0$

$$\boxed{F'(0) < -\frac{mg}{2}}$$

Problem 5

(a)

$$T = \frac{1}{2} m (\dot{s}^2 + s^2 \cos^2 \alpha \dot{\theta}^2) + \frac{1}{2} m \dot{s}^2$$
$$= m \left(\dot{s}^2 + \frac{1}{2} s^2 \cos^2 \alpha \dot{\theta}^2 \right)$$

$$V = mg s \sin \alpha + mgs = (1 + \sin \alpha) mgs$$

$$L = m \left(\dot{s}^2 + \frac{\cos^2 \alpha}{2} s^2 \dot{\theta}^2 \right) - (1 + \sin \alpha) mgs$$

(b)

θ is ignorable

$$\frac{\partial L}{\partial \dot{\theta}} = C_{\theta}; \quad m \cos^2 \alpha s^2 \dot{\theta} = C_{\theta}$$

(conservation of angular momentum)

(c)

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{s}} \right) = 2m\ddot{s}$$

$$\frac{\partial L}{\partial s} = m \cos^2 \alpha s \dot{\theta}^2 - (1 + \sin \alpha) mg$$

$$\dot{\theta} = \frac{C_{\theta}}{m \cos^2 \alpha s^2}$$

Equation of motion:

$$2m\ddot{s} - m\cos^2\alpha s \frac{C_\theta^2}{m^2\cos^4\alpha s^3} + (1+\sin\alpha)mg = 0$$

$$\ddot{s} - \frac{C_\theta^2}{2m^2\cos^2\alpha s^3} + \frac{(1+\sin\alpha)g}{2} = 0$$

Steady motion:

$$\ddot{s} = \dot{s} = 0$$

$$C_\theta = m\cos^2\alpha s^2 \dot{\theta}$$

$$\frac{C_\theta^2}{2m^2\cos^2\alpha s^3} = \frac{(1+\sin\alpha)g}{2}$$

$$\frac{m^2\cos^4\alpha s^3 \dot{\theta}^2}{2m^2\cos^2\alpha s^3} = \frac{(1+\sin\alpha)g}{2}$$

$$\cos^2\alpha s \dot{\theta}^2 = (1+\sin\alpha)g$$

(d)

Convert to first order:

$$\dot{s} = v$$

$$\dot{v} = \frac{C_\theta^2}{2m^2\cos^2\alpha s^3} - \frac{(1+\sin\alpha)g}{2}$$

Linearize:

$$\begin{bmatrix} \dot{s} \\ \dot{r} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{-3C_0^2}{2m^2 \cos^2 \alpha S^4} & 0 \end{bmatrix} \begin{bmatrix} s \\ r \end{bmatrix}$$

$$S\theta^2 = \frac{1 + \sin \alpha}{\cos^2 \alpha} g$$

Eigenvalues:

$$\begin{vmatrix} -\lambda & 1 \\ \frac{-3C_0^2}{2m^2 \cos^2 \alpha S^4} & -\lambda \end{vmatrix} = \lambda^2 + \frac{3C_0^2}{2m^2 \cos^2 \alpha S^4} = 0$$

$$S\theta^2 = \frac{1 + \sin \alpha}{\cos^2 \alpha} g$$

Since $\frac{3C_0^2}{2m^2 \cos^2 \alpha S^4} > 0$ always,

$\lambda \in \text{Im} \Rightarrow \text{Stable}$