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1

$$\frac{dh}{dt} = \frac{d}{dt} \left(\dot{q} \frac{\partial L}{\partial \dot{q}} - L \right) = \ddot{q} \frac{\partial L}{\partial \dot{q}} + \dot{q} \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{dL}{dt} =$$

$$= \ddot{q} \frac{\partial L}{\partial \dot{q}} + \dot{q} \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{\partial L}{\partial \dot{q}} \cdot \ddot{q} - \frac{\partial L}{\partial q} \cdot \dot{q} =$$

$$= \underbrace{\ddot{q} \left(\frac{\partial L}{\partial \dot{q}} - \frac{\partial L}{\partial q} \right)}_{=0} + \dot{q} \underbrace{\left(\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}} \right) - \frac{\partial L}{\partial q} \right)}_{=0} =$$

= 0 q.e.d.

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(2)

a. The potential energy of a system must attain a strict relative minimum on a stable equilibrium configuration.

b.

$$\begin{aligned} V &= -mg 2l \sin\theta + mg l \cos 2\theta \\ &= mg l (\cos 2\theta - 2 \sin\theta) \end{aligned}$$

Equilibrium:

$$\frac{dV}{d\theta} = 0 \Rightarrow mg l (-2 \sin 2\theta - 2 \cos\theta) = 0$$

$$2 \sin\theta \cos\theta + \sin\theta = 0$$

$$\sin\theta (2 \sin\theta + 1) = 0$$

$$\sin\theta = 0 \Rightarrow \theta = \pm 90^\circ$$

$$\sin\theta = -\frac{1}{2} \Rightarrow \theta = \begin{cases} 210^\circ \\ 330^\circ \end{cases}$$

Stability:

$$\frac{d^2V}{d\theta^2} = (-4 \sin 2\theta + 2 \cos\theta) mg l$$

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$$\theta = +90^\circ: \frac{d^2V}{d\theta^2} = (-4 \cos(180) + 2 \sin(90)) \text{mgl} = \\ = 6 \text{mgl} > 0 \Rightarrow \text{minimum}$$

STABLE

$$\theta = -90^\circ: \frac{d^2V}{d\theta^2} = (-4 \cos(-180) + 2 \sin(-90)) \text{mgl} = \\ = 2 \text{mgl} > 0 \Rightarrow \text{minimum}$$

STABLE

$$\theta = 210^\circ: \frac{d^2V}{d\theta^2} = (-4 \cos(520) + 2 \sin(210)) \text{mgl} = \\ = -3 \text{mgl} < 0 \Rightarrow \text{maximum}$$

UNSTABLE

$$\theta = 330^\circ: \frac{d^2V}{d\theta^2} = (-4 \cos(660) + 2 \sin(330)) \text{mgl} = \\ = -3 \text{mgl} < 0 \Rightarrow \text{maximum}$$

UNSTABLE

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(3) a.

$$I_{xy} = I_{xz} = I_{yz} = 0 \quad \text{because of symmetry}$$

$$I_{xx} = \frac{1}{3} ml^2 + 4ml^2 = \frac{13}{3} ml^2$$

$$I_{yy} = 2 \cdot \frac{1}{3} (2m)(2l)^2 = \frac{16}{3} ml^2$$

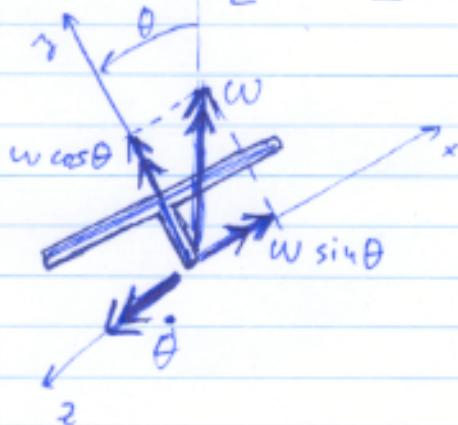
$$I_{zz} = \underbrace{I_{xx} + I_{yy}}_{\text{because it is a flat body in } xy} = \frac{29}{3} ml^2$$

because it is a flat body in xy

$$\underline{\underline{I}_o} = \begin{bmatrix} \frac{13}{3} & 0 & 0 \\ 0 & \frac{16}{3} & 0 \\ 0 & 0 & \frac{29}{3} \end{bmatrix} ml^2$$

b.

$$T = \frac{1}{2} \underline{\underline{w}}^T \underline{\underline{I}_o} \underline{w}$$



$$\underline{w} = \begin{pmatrix} w \sin \theta \\ w \cos \theta \\ \dot{\theta} \end{pmatrix}$$

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$$T = \frac{1}{2} ml^2 [w \sin \theta \quad w \cos \theta \quad \dot{\theta}]$$

$$\begin{bmatrix} \frac{13}{3} & 0 & 0 \\ 0 & \frac{16}{3} & 0 \\ 0 & 0 & \frac{29}{3} \end{bmatrix} \begin{bmatrix} w \sin \theta \\ w \cos \theta \\ \dot{\theta} \end{bmatrix}$$

$$= \frac{1}{2} ml^2 \left(\frac{13}{3} w^2 \sin^2 \theta + \frac{16}{3} w^2 \cos^2 \theta + \frac{29}{3} \dot{\theta}^2 \right) =$$

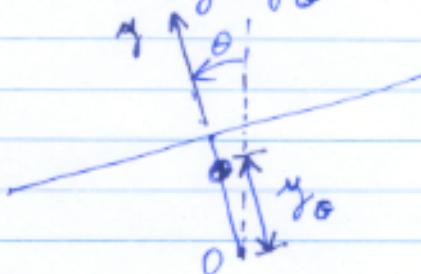
$$= \frac{1}{2} ml^2 \left(\frac{13}{3} w^2 + w^2 \cos^2 \theta + \frac{29}{3} \dot{\theta}^2 \right)$$

$$= ml^2 \left[\frac{29}{6} \dot{\theta}^2 + \left(\frac{13}{6} + \frac{1}{2} \cos^2 \theta \right) w^2 \right]$$

c.

$$L = T - V$$

$$V = 5mg y_0 \cos \theta$$



$$y_0 = \left(4ml + m \frac{l}{2} \right) / 5m$$

$$= \frac{9l}{10}$$

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$$V = 5mg \frac{9l}{10} \cos\theta = \frac{9}{2} mgl \cos\theta$$

$$L = ml^2 \left[\frac{29}{6} \dot{\theta}^2 + \left(\frac{13}{6} + \frac{1}{2} \cos^2\theta \right) w^2 \right] - \frac{9}{2} mgl \cos\theta$$

Equation of motion:

$$\frac{\partial L}{\partial \dot{\theta}} = \frac{29}{3} ml^2 \ddot{\theta} \quad \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) = \frac{29}{3} ml^2 \ddot{\theta}$$

$$\frac{\partial L}{\partial \theta} = -ml^2 \sin\theta \cos\theta w^2 + \frac{9}{2} mgl \sin\theta$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{\theta}} \right) - \frac{\partial L}{\partial \theta} = 0$$

$$\frac{29}{3} ml^2 \ddot{\theta} + ml^2 \sin\theta \cos\theta w^2 - \frac{9}{2} mgl \sin\theta = 0$$

$$\boxed{\frac{29}{3} \ddot{\theta} + \frac{1}{2} (\sin 2\theta) w^2 - \frac{9}{2} \frac{g}{l} \sin\theta = 0}$$

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d. For $\theta = 0$ $\ddot{\theta} = 0$, thus $\theta = 0$ is an equilibrium position indeed.

Stability:

Equation of motion was found in c,
stability analysis by linearisation is
convenient.

$$\dot{\theta} = \varphi$$

$$\dot{\varphi} = \frac{3}{29} \left(-\frac{1}{2}(\sin 2\theta) w^2 + \frac{9g}{2l} \sin \theta \right)$$

Linearisation for $\theta = 0$:

$$\begin{bmatrix} \dot{\theta} \\ \dot{\varphi} \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ \frac{3}{29} \left(-(\cos 2\theta) w^2 + \frac{9g}{2l} \cos \theta \right) & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \varphi \end{bmatrix}$$

$$= \begin{bmatrix} 0 & 1 \\ \frac{3}{29} \left(\frac{9g}{2l} - w^2 \right) & 0 \end{bmatrix} \begin{bmatrix} \theta \\ \varphi \end{bmatrix}$$

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$$\begin{vmatrix} -\lambda & 1 \\ \frac{3}{2l} \left(\frac{q_2}{2e} - w^2 \right) & -\lambda \end{vmatrix} = \lambda^2 - \frac{3}{2l} \left(\frac{q_2}{2e} - w^2 \right) = 0$$

$$\lambda \in \text{Im} \Leftrightarrow \frac{q_2}{2l} - w^2 < 0$$

$$\text{Stable Equilibrium} \Leftrightarrow w^2 > \frac{q_2}{2l}$$

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(4)

$$\omega_1 = \underline{\omega}_1, \quad \omega_2 = \underline{\omega}_2, \quad \omega_3 = \underline{\omega}_3 + p$$

$$\dot{\omega}_1 = \underline{\dot{\omega}}_1, \quad \dot{\omega}_2 = \underline{\dot{\omega}}_2, \quad \dot{\omega}_3 = \underline{\dot{\omega}}_3 + \dot{p}$$

$$\dot{p} = 0$$

$$0 = \frac{d\underline{\omega}}{dt} = \underline{\dot{\omega}}_1 \underline{i} + \underline{\dot{\omega}}_2 \underline{j} + \underline{\dot{\omega}}_3 \underline{k} + \underline{\omega} \times \underline{\omega}$$

$$= \underline{\omega}_1 \underline{i} + \underline{\omega}_2 \underline{j} + \underline{\omega}_3 \underline{k} + \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ \underline{\omega}_1 & \underline{\omega}_2 & \underline{\omega}_3 + p \\ \underline{\omega}_1 & \underline{\omega}_2 & \underline{\omega}_3 \end{vmatrix}$$

$$= (\underline{\omega}_1 - \underline{\omega}_3 p) \underline{i} + (\underline{\omega}_2 + \underline{\omega}_3 p) \underline{j} + \underline{\omega}_3 \underline{k} = 0$$

$$\underline{\omega}_1 = \underline{\omega}_3 p \quad \underline{\omega}_2 = -\underline{\omega}_3 p \quad \underline{\omega}_3 = 0$$

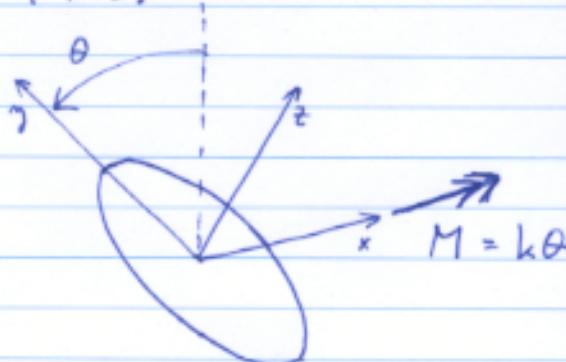
Consequently

$$\dot{\omega}_1 = -\underline{\omega}_3 p \quad \dot{\omega}_2 = -\underline{\omega}_3 p \quad \dot{\omega}_3 = 0$$

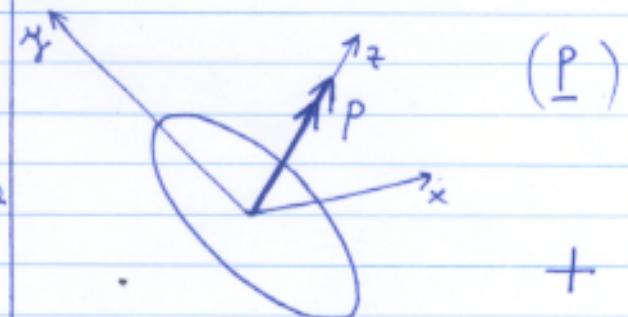
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b.

FBD:

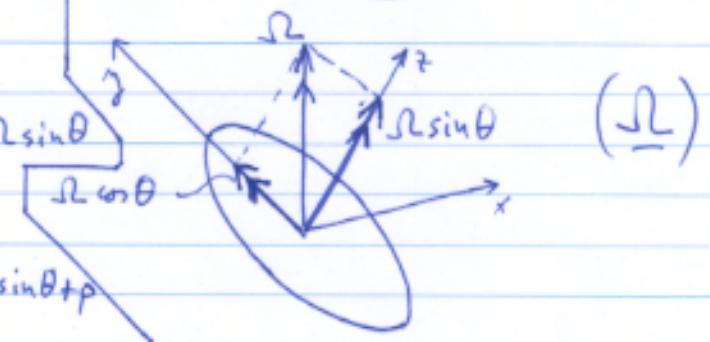


ANGULAR VELOCITIES:



+

$$\Omega_1 = 0; \quad \Omega_2 = \Omega \cos \theta; \quad \Omega_3 = \Omega \sin \theta$$



$$\dot{\omega}_1 = \Omega \cos \theta p; \quad \dot{\omega}_2 = 0; \quad \dot{\omega}_3 = 0$$

$$M_1 = k\theta; \quad M_2 = 0; \quad M_3 = 0; \quad I_1 = I_2 = \frac{1}{4}mr^2; \quad I_3 = \frac{1}{2}mr^2$$

$$\left. \begin{aligned} -I_1 \ddot{\omega}_1 - (I_2 - I_3) \omega_2 \omega_3 &= M_1 \\ I_2 \ddot{\omega}_2 - (I_3 - I_1) \omega_1 \omega_3 &= M_2 \\ I_3 \ddot{\omega}_3 - (I_1 - I_2) \omega_1 \omega_2 &= M_3 \end{aligned} \right\} \begin{aligned} 0 &= 0 \\ 0 &= 0 \\ 0 &= 0 \end{aligned}$$

$$\rightarrow \frac{1}{4}mr^2 \ddot{\omega} \cos \theta p - \left[\left(\frac{1}{4} - \frac{1}{2} \right) mr^2 \right] \ddot{\omega} \cos \theta (\omega \sin \theta + p) = k\theta$$

$$\frac{1}{2}mr^2 \ddot{\omega} \cos \theta p + \underbrace{\frac{1}{4}mr^2 \ddot{\omega}^2 \sin \theta \cos \theta}_{\text{neglected because of } \ddot{\omega}^2} = k\theta$$

neglected because of $\ddot{\omega}^2$

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$$\frac{1}{2} m R^2 \Omega \sin \theta p = k \theta$$

$$\Omega = \frac{k \theta}{\frac{1}{2} m R^2 p \sin \theta}$$

c.

$$m = 0,2 \text{ kg} \quad R = 0,05 \text{ m} \quad p = 1000 \text{ s}^{-1}$$

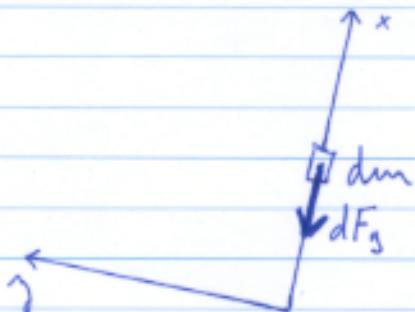
$$k = 6,25 \times 10^{-3} \text{ Nm} \quad \text{and} \quad \theta = 60^\circ$$

$$\Rightarrow \Omega = 3^\circ \text{ s}^{-1} \quad (\text{standard turn!})$$

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(5)

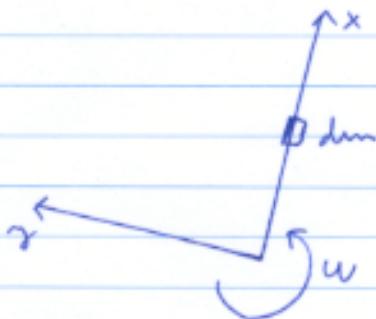
a. Actual forces: Gravity



$$\underline{dF_g} = -dm \cdot g \underline{i}$$

$$= -\rho A \frac{k}{x^2} \underline{i} dx$$

Fictitious forces: Centrifugal force



$$\underline{w} = w \underline{k}$$

$$\underline{v}_{\text{rel}} = \underline{x} \underline{i}$$

$$\underline{N}_{\text{rel}} = 0$$

$$\underline{dF_{\text{fict}}} = -dm \left(\underbrace{\underline{w} \times \underline{v}_{\text{rel}} + \underline{w} \times (\underline{w} \times \underline{v}_{\text{rel}})}_{\underline{w} \times (\underline{w} \times \underline{v}_{\text{rel}})} + 2(\underline{w} \times \underline{N}_{\text{rel}}) \right)$$

$$\underline{w} \times (\underline{w} \times \underline{v}_{\text{rel}}) = \underline{w} \times \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ 0 & 0 & w \\ x & 0 & 0 \end{vmatrix} = \underline{w} \times (\underline{w} \times \underline{j})$$

$$= \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ 0 & 0 & w \\ 0 & w & 0 \end{vmatrix} = -w^2 \underline{x} \underline{i}$$

$$\underline{dF_{\text{fict}}} = + dm w^2 \underline{x} \underline{i} = \rho A w^2 \underline{x} \underline{i} dx$$

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Total forces (as measured on Earth-tower system)

$$dF = dF_g + dF_{\text{free}} = \left(-\rho A \frac{k}{x^2} + \rho A \omega^2 x \right) dx \hat{i}$$

which is in the x -direction.

$$dF(x) = dF_x = \rho A \left(\omega^2 x - \frac{k}{x^2} \right) dx$$

- b. The generalised coordinate for the motion of a point of the tower is $\mathbf{u}(x, t)$.

For a differential of mass one has

$$-\frac{\partial}{\partial u} (dV_{\text{gen}}) = dF$$

dF does not depend on $\mathbf{u}(x, t)$, so we can write:

$$dV_{\text{gen}} = -dF \cdot \mathbf{u}$$

in other words

$$dV_{\text{gen}} = -\rho A \left(\omega^2 x - \frac{k}{x^2} \right) \mathbf{u} dx$$

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c.

$$dL = dT - dV$$

$$dT = \frac{1}{2} \nu_{\text{rel}}^2 dm = \frac{1}{2} \rho A u_t^2 dx \quad \left[M_t = \frac{\partial M}{\partial t} \right]$$

$$dV = dV_{\text{el}} + dV_{\text{gen}}$$

$$dV_{\text{el}} = \frac{1}{2} EA \varepsilon^2 dx = \frac{1}{2} EA u_x^2 dx \quad \left[M_x = \frac{\partial M}{\partial x} \right]$$

$$dV = \left[\frac{1}{2} EAM_x^2 - \rho A \left(u_x^2 - \frac{k}{x^2} \right) M \right] dx$$

$$dL = \left[\frac{1}{2} \rho A u_t^2 - \frac{1}{2} EA u_x^2 + \rho A \left(u_x^2 - \frac{k}{x^2} \right) M \right] dx$$

d.

The basis of the tower is fixed to the Earth, which is considered to be rigid. Consequently,

$$M(R_e, t) = 0$$

which implies $\delta M(R_e, t) = 0$.

This is an ESSENTIAL boundary condition

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e.

$$\begin{aligned}
 I(\mu) &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} dL dt \\
 &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} \left[\frac{1}{2} \rho A \dot{M}_e^2 - \frac{1}{2} E A M_e^2 + \rho A \left(\dot{w}_x - \frac{k}{x^2} \right) \mu \right] dx dt \\
 &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} \phi(x, \mu, \dot{w}_x, M_e) dx dt
 \end{aligned}$$

$$\begin{aligned}
 \delta I(\mu) &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} \delta \phi(x, \mu, \dot{w}_x, M_e) dx dt = \\
 &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} \left[\frac{\partial \phi}{\partial \mu} \delta \mu + \frac{\partial \phi}{\partial \dot{w}_x} \delta \dot{w}_x + \frac{\partial \phi}{\partial M_e} \delta M_e \right] dx dt = \\
 &= \int_{t_a}^{t_b} \int_{R_e}^{R_g} \left[\frac{\partial \phi}{\partial \mu} - \frac{\partial}{\partial x} \left(\frac{\partial \phi}{\partial \dot{w}_x} \right) - \frac{\partial}{\partial t} \left(\frac{\partial \phi}{\partial M_e} \right) \right] \delta \mu dx dt +
 \end{aligned}$$

$$\begin{aligned}
 &\stackrel{(1)}{\rightarrow} + \int_{t_a}^{t_b} \int_{R_e}^{R_g} \left. \frac{\partial \phi}{\partial \mu} \delta \mu \right|_{K_2} dt + \int_{R_e}^{R_g} \left. \frac{\partial \phi}{\partial \mu} \delta \mu \right|_{t_a}^{t_b} dx \\
 &\quad \underbrace{\qquad\qquad\qquad}_{(2)} \qquad\qquad\qquad \underbrace{\qquad\qquad\qquad}_{(3)}
 \end{aligned}$$

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① = 0 for any variation $\delta M \Rightarrow$

$$\frac{\partial \phi}{\partial M} - \frac{\partial}{\partial x} \left(\frac{\partial \phi}{\partial M_x} \right) - \frac{\partial}{\partial t} \left(\frac{\partial \phi}{\partial M_t} \right) = 0$$

Substituting ϕ :

$$\rho A M_{tt} - E A M_{xx} = \rho A \left(w^2 x - \frac{k}{x^2} \right)$$

$$\boxed{\rho M_{tt} - E M_{xx} = \rho \left(w^2 x - \frac{k}{x^2} \right)} \quad \text{Equation of motion}$$

$$② = 0 \Rightarrow 0 = \int_{t_a}^{t_b} \left[\frac{\partial \phi}{\partial M_x} \Big|_{x=R_3} - \frac{\partial \phi}{\partial M_x} \Big|_{x=R_e} \right] dt$$

\downarrow \downarrow

$\neq 0$, free see d.

$$\frac{\partial \phi}{\partial M_x} \Big|_{x=R_3} = 0 \Rightarrow E A M_x(R_3, t) = 0$$

$$\boxed{M_x(R_3, t) = 0}$$

This is a NATURAL boundary condition

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(3) = 0 would provide natural boundary conditions for $u(x, t_1)$ and $u(x, t_2)$, which are not asked for.

f. Additional fictitious force:

$$dF_{\text{fict}} = -dm \left(\underline{\dot{w}} \times \underline{r}_{\text{rel}} + \underline{w} \times (\underline{w} \times \underline{r}_{\text{rel}}) + 2(\underline{w} \times \underline{N}_{\text{rel}}) \right)$$

The component coming from $\underline{w} \times (\underline{w} \times \underline{r}_{\text{rel}})$ was already found, and $\underline{\dot{w}} = 0$.

Now we also have $\underline{N}_{\text{rel}} = M_t \underline{i}$, so

$$2 \underline{w} \times \underline{N}_{\text{rel}} = 2 \begin{vmatrix} \underline{i} & \underline{j} & \underline{k} \\ 0 & 0 & w \\ M_t & 0 & 0 \end{vmatrix} = \\ = 2 w M_t \underline{j}$$

thus $dF_{\text{fict}} = -dm(-w^2 \times \underline{i} + 2 w M_t \underline{j})$

so the additional term is

$$\boxed{-\rho A 2 w M_t dx}$$

which is a Coriolis force, in the y -direction