## State space models

AE4-E08 Satellite Navigation

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## State vector description

## Dynamic systems

first order vector differential equation (state equation)

ullet general form (possibly non-linear, and dependent on time t)

$$\dot{x}(t) = f(x(t), z(t), t)$$
  
state-vector input

linear time-varying

$$\dot{x}(t) = F(t)x(t) + G(t)z(t)$$

linear time-invariant

$$\dot{x}(t) = Fx(t) + Gz(t)$$

## Linear time-invariant state equation

$$\dot{x}(t) = Fx(t) + Gz(t)$$

with 
$$\dot{\Phi}(t,t_0) = F\Phi(t,t_0)$$

$$\Phi(t,t_0) = e^{F(t-t_0)}$$
 where  $e^{Ft} = \sum_{i=0}^{\infty} \frac{F^i t^i}{i!}$  (matrix exponential)

full solution

$$x(t) = \underbrace{e^{F(t-t_0)}}_{t_0} x(t_0) + \int_{t_0}^t \underbrace{e^{F(t-\tau)}}_{t_0} Gz(\tau) d\tau$$

$$\Phi(t,t_0)$$