

Classical Plate theory:-



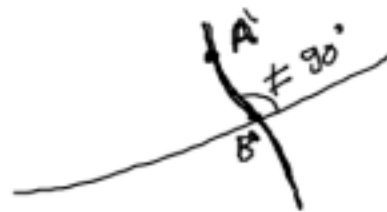
mid surface: $\bar{u}(x, y)$ $\bar{v}(x, y)$ $\bar{w}(x, y)$

(x, y, z) undeformed $!?$ $(x+u, y+v, z+w)$ deformed

(x, y, z) connected $(x, y, 0)$



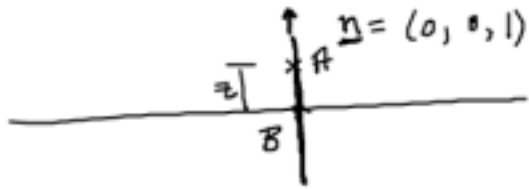
Kirchhoff



deformed

- 1- stretch or shrink X
- 2- it may deform X
- 3- it may not remain normal to the surface. X

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$$A: (x, y, z)$$

$$(x, y, w) + z \underline{n}' : A'$$

$$B: (x, y, 0) \longrightarrow (x, y, w) : B'$$

$$\bar{w}(x, y), \quad \underline{n}'(x, y)$$

$$(1, 0, \bar{w}_x), \quad (0, 1, \bar{w}_y)$$

$$\underline{n}' = \frac{(-\bar{w}_x, -\bar{w}_y, 1)}{\sqrt{1 + \bar{w}_x^2 + \bar{w}_y^2}}$$

$$\underline{n}' \approx (-\bar{w}_x, -\bar{w}_y, 1)$$

$$(x, y, z) \longrightarrow (x - z\bar{w}_x, y - z\bar{w}_y, z + w)$$

$$(x, y, z) \longrightarrow (x - z w_{,x}, y - z w_{,y}, z + w)$$

$$\left. \begin{aligned} u &= -z w_{,x} \\ v &= -z w_{,y} \end{aligned} \right\}$$

$$\boxed{w = w(x, y)}$$

Displacements

Strains:

$$\epsilon_x = -z w_{,xx}$$

$$\epsilon_z = 0$$

$$\epsilon_y = -z w_{,yy}$$

$$\begin{aligned} \gamma_{xz} &= -w_{,x} + x w_{,x} \\ &= 0 \end{aligned}$$

$$\gamma_{xy} = -2z w_{,xy}$$

$$\begin{aligned} \gamma_{yz} &= -w_{,y} + y w_{,y} \\ &= 0 \end{aligned}$$

* state of plane stress

$$\tau_{xz} = 0 \quad \tau_{yz} = 0$$

$$\sigma_z = 0$$



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$$W = - \iiint \sigma_x \delta \epsilon_x + \sigma_y \delta \epsilon_y + \tau_{xy} \delta \gamma_{xy} \, dx \, dy \, dz$$

$$\epsilon_x = +z (\nu_{,xx}) \quad \sigma_x = \frac{E}{1-\nu^2} (-w_{,xx} - \nu w_{,yy}) z$$

$$\epsilon_y = z (-w_{,yy}) \quad \sigma_y = \frac{E}{1-\nu^2} (-w_{,yy} - \nu w_{,xx}) z$$

$$\gamma_{xy} = z (-2w_{,xy}) \quad \tau_{xy} = \frac{E}{2(1+\nu)} (-2w_{,xy}) z$$

$$\underline{(M_x, M_y, M_{yx})}$$

$$(Q_x, Q_y)$$

$$Q_{y,x} + Q_{x,y} + q = 0$$

$$M_x = \int_{-t/2}^{t/2} \sigma_x z \, dz = \frac{E}{1-\nu^2} (-w_{,xx} - \nu w_{,yy}) \int_{-t/2}^{t/2} z^2 \, dz$$

$$M_x = \frac{Et^3}{12(1-\nu^2)} (-w_{,xx} - \nu w_{,yy})$$

D: bending stiffness

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$$M_x = D(-w_{,xx} - \nu w_{,yy})$$

$$M_y = D(-\nu w_{,xx} - w_{,yy})$$

$$M_{yx} = D \frac{(1-\nu)}{2} (-2w_{,xy})$$

$$\begin{pmatrix} M_x \\ M_y \\ M_{yx} \end{pmatrix} = D \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{pmatrix} -w_{,xx} \\ -w_{,yy} \\ -2w_{,xy} \end{pmatrix}$$

$$Q_{x,x} + Q_{y,y} + q = 0$$

$$M_{y,x} + M_{x,y} = Q_x$$

$$M_{x,y} + M_{y,x} = Q_y$$

$$M_{x,xx} + 2M_{y,x,y} + M_{y,yy} + q = 0$$

$$\boxed{-D[w_{,xxxx} + 2w_{,xxyy} + w_{,yyyy}] + q = 0}$$

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$$\left. \begin{aligned} \sigma_x &= \frac{12}{t^3} M_x z \\ \sigma_y &= \frac{12}{t^3} M_y z \\ \tau_{xy} &= \frac{12}{t^3} M_{xy} z \end{aligned} \right\}$$

Solution procedure:

- 1) Solve the Classical Plate equation for $w(x, y)$
- 2) Calculate (M_x, M_y, M_{xy})
- 3) Calculate stress dist. $(\sigma_x, \sigma_y, \tau_{xy})$