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Stress - strain relation

Constitutive Law (relation)

$$\underline{\sigma} = \frac{\partial \psi(\underline{\epsilon})}{\partial \underline{\epsilon}}$$

$$\begin{aligned} \sigma_x &= \frac{\partial \psi}{\partial \epsilon_x} & \sigma_y &= \frac{\partial \psi}{\partial \epsilon_y} \\ \frac{\partial \sigma_x}{\partial \epsilon_y} &= \frac{\partial^2 \psi}{\partial \epsilon_y \partial \epsilon_x} & \frac{\partial \sigma_y}{\partial \epsilon_x} &= \frac{\partial^2 \psi}{\partial \epsilon_x \partial \epsilon_y} \\ & & &= \\ & \left(\frac{\partial \sigma_x}{\partial \epsilon_y} = \frac{\partial \sigma_y}{\partial \epsilon_x} \right) \end{aligned}$$

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Case of zero initial stress

$$0 = \left. \frac{\partial \psi}{\partial \varepsilon_x} \right|_{\underline{\varepsilon}=\underline{0}} \quad 0 = \left. \frac{\partial \psi}{\partial \varepsilon_y} \right|_{\underline{\varepsilon}=\underline{0}} \quad \dots$$

$$\psi(\underline{\varepsilon}=\underline{0}) = 0$$

$$\psi(\underline{\varepsilon}) = \underbrace{\psi(\underline{0})}_0 + \underbrace{\left. \frac{\partial \psi}{\partial \underline{\varepsilon}} \right|_{\underline{\varepsilon}=\underline{0}}}_{\underline{0}} \underline{\varepsilon} + \frac{1}{2} \underline{\varepsilon}^t \underline{\underline{C}} \underline{\varepsilon} + \text{H.O.T}$$

$$\psi(\underline{\varepsilon}) \sim \frac{1}{2} \underline{\varepsilon}^t \underline{\underline{C}} \underline{\varepsilon}$$

$\underline{\underline{C}}$: is a 6x6 symmetric matrix

$\underline{\underline{C}}$: the material stiffness matrix
elasticity matrix

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$$\underline{\sigma} = \underline{C} \underline{\varepsilon}$$

$$\Psi = \frac{1}{2} \underline{\varepsilon}^t \underline{C} \underline{\varepsilon}$$

$$\Psi > 0 \quad \forall \underline{\varepsilon} \neq \underline{0}$$

C is symmetric & positive definite

21 elasticities

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Isotropic material

$$\underline{\underline{\sigma}} = \underline{\underline{T}}_{\sigma} \underline{\underline{\sigma}}'$$

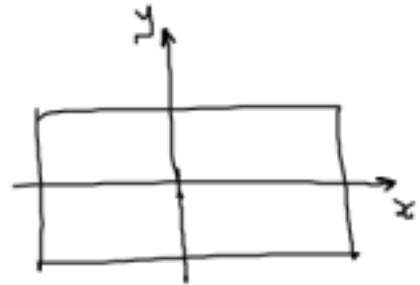
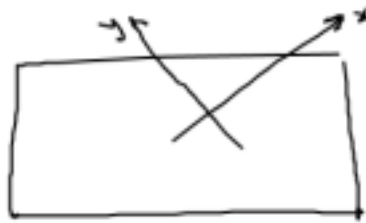
$$\underline{\underline{\varepsilon}} = \underline{\underline{T}}_{\varepsilon} \underline{\underline{\varepsilon}}'$$

$$\underline{\underline{\sigma}} = \underline{\underline{C}} \underline{\underline{\varepsilon}}$$

$$\underline{\underline{T}}_{\sigma} \underline{\underline{\sigma}}' = \underline{\underline{C}} \underline{\underline{T}}_{\varepsilon} \underline{\underline{\varepsilon}}'$$

$$\underline{\underline{\sigma}}' = (\underline{\underline{T}}_{\sigma}^{-1} \underline{\underline{C}} \underline{\underline{T}}_{\varepsilon}) \underline{\underline{\varepsilon}}'$$

$$\underline{\underline{C}}' = \underline{\underline{T}}_{\sigma}^{-1} \underline{\underline{C}} \underline{\underline{T}}_{\varepsilon} \neq$$



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$$\underline{\underline{C'}} = \underline{\underline{C}} \quad (\text{condition of isotropy})$$

$$\left. \begin{array}{l} \text{Young's modulus } (E) \\ \text{Poisson's ratio } (\nu) \end{array} \right\} \Rightarrow \text{Shear modulus } (G)$$

$$G = \frac{E}{2(1+\nu)}$$

$$\varepsilon_x = \frac{\sigma_x}{E} + \left(-\nu \frac{\sigma_y}{E}\right) + \left(-\nu \frac{\sigma_z}{E}\right)$$

$$\varepsilon_y = -\nu \frac{\sigma_x}{E} + \frac{\sigma_y}{E} + \left(-\nu \frac{\sigma_z}{E}\right)$$

$$\varepsilon_z = -\nu \frac{\sigma_x}{E} + \left(-\nu \frac{\sigma_y}{E}\right) + \frac{\sigma_z}{E}$$

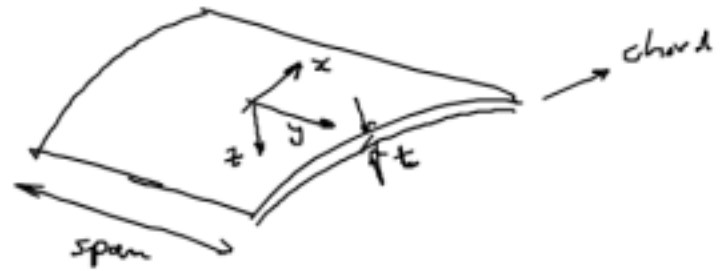
$$\gamma_{yz} = \frac{\tau_{yz}}{G} \quad \gamma_{zx} = \frac{\tau_{zx}}{G} \quad \gamma_{xy} = \frac{\tau_{xy}}{G}$$

Plane stress :-

$$\underline{\sigma_z = 0} \quad \tau_{xz} = 0 \quad \tau_{yz} = 0$$

$$\gamma_{xz} = \frac{\tau_{xz}}{G} = 0$$

$$\gamma_{yz} = \frac{\tau_{yz}}{G} = 0$$



$$\left. \begin{aligned} \epsilon_x &= \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} \\ \epsilon_y &= -\nu \frac{\sigma_x}{E} + \frac{\sigma_y}{E} \\ \gamma_{xy} &= \frac{\tau_{xy}}{G} \end{aligned} \right\}$$

$$\epsilon_z \neq 0$$

$$\begin{pmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \end{pmatrix} = \frac{E}{1-\nu^2} \begin{pmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{pmatrix} \begin{pmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{pmatrix}$$

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Strain energy expression for plane stress:

$$W_{in} = - \int (\sigma_x \delta \epsilon_x + \sigma_y \delta \epsilon_y + \tau_{xy} \delta \gamma_{xy} + \sigma_z \delta \epsilon_z + \tau_{yz} \delta \gamma_{yz} + \tau_{zx} \delta \gamma_{zx}) dV$$

$$W_{in} = - \int \sigma_x \delta \epsilon_x + \sigma_y \delta \epsilon_y + \tau_{xy} \delta \gamma_{xy} dV$$

$$W_{in} = - \int \delta \psi dV$$

$$\psi = \frac{1}{2} \int \frac{E}{1-\nu^2} \left(\epsilon_x^2 + \epsilon_y^2 + 2\nu \epsilon_x \epsilon_y + \left(\frac{1-\nu}{2}\right) \gamma_{xy}^2 \right) dV$$